SA Based Software Deployment Reliability Estimation Considering Component Reliability of Exponential Distribution

Xihong Su School of Computer Science and Technology, Harbin Institute of Technology, Harbin, China Email: suxihong07@gmail.com

Hongwei Liu, Zhibo Wu, Xiaozong Yang, Decheng Zuo School of Computer Science and Technology, Harbin Institute of Technology, Harbin, China Email: {lhw, wzb, xzy, zdc}@ftcl.hit.edu.cn

Abstract-Although many approaches for architecturebased reliability estimation exist, these approaches are typically limited to certain classes or exclusively concentrate on software reliability, neglecting the influence of hardware resources, component reliability, component replica and software deployment. In this paper, a reliability estimation model based on software architecture (SA) is presented. This model incorporates the influence of software deployment, component reliability and component replica. Component lifetimes can be modeled by exponential distribution. The approach of calculating system reliability considering component replica and component reliability is proposed. The influences of different deployment architectures on component reliabilities and system reliability are investigated. The improvement of system reliability by redeployment or component replica is discussed.

Index Terms—software architecture, software deployment, component reliability, exponential distribution, component replica, system redeployment

I. INTRODUCTION

The past few decades have witnessed an unrelenting pattern of growth in size and complexity of software systems, which will likely continue well into the foreseeable future. This pattern is further evident in an emerging class of embedded and pervasive software systems that are growing in popularity due to increase in the speed and capacity of hardware, decrease in its cost, emergency of wireless ad hoc networks, proliferation of sensors and handheld computing devise, etc. Studies have shown that a promising approach to resolve the challenges of developing large scale software system is to employ the principles of software architectures [1,2]. Software architecture provides abstractions for representing the structure, behavior, and key properties of a software system. They are described in terms of components (computational software elements). (interaction elements), and their connectors configurations [3, 4].

Many approaches have begun to predict reliability at the level of architectural models, or at least in terms of high-level system structure [5-12]. Firstly, these researchers acknowledge that reliabilities of components have a significant impact on system reliability, but they almost invariably assume that the reliabilities of the components in a system are known. The few researchers consider component-level reliability [8, 10], assume that the reliabilities of a given component elements, such as its services, are known. Secondly, these approaches are typically limited to certain failure classes or exclusively concentrate on software reliability, neglecting the influence of hardware resources, software deployment and component replica. In a given context, some of these deployment configurations are obviously more effective than others in terms of reliability. Additionally, if replica of critical software components exists, failure of one host node does not mean that the whole system fails. Therefore, we propose a new SA based reliability estimation model, incorporating the influence of software deployment, component replica and component reliability.

The rest of this paper is organized as follows. Section II describes system deployment architecture. Section III investigates component replica and component reliability. Section IV proposes an approach of calculating system reliability. Section V gives the experiments. Section VI presents the conclusions and directions of future study.

II. SYSTEM DEPLOYMENT ARCHITECTURE

In this section, we present an overview of whole structure of system deployment architecture. System

Manuscript received September 8, 2010; accepted February 18, 2011.

Supported by High Technology Research and Development Program of China (Project No. 2008AA01A201), National High Technology Research and Development Plan of China (Project No. 2006AA01A103), National Nature Science Funds of China (Project No. 60503015)

Communication author: Xihong Su, born in 1981, female, Ph.D. Harbin Institute of Technology, Harbin 150001, China

deployment architecture is the allocation of the system software components on its hardware host nodes [13]. We also introduce the frequency matrix CC of interaction among software components.

A. Whole Structure

The basic entities of SA based software deployment reliability estimation model include host nodes, software components and services. In details, the model consists of

1) a set *H* of host nodes, $H = \{H_1, H_2, \dots, H_m\}$, which represents the host nodes of a system.

2) a set C of components, $C = \{C_1, C_2, \dots, C_n\}$. Each component may have multiple component replica.

3) a set S of services, which describes the different use cases that the whole system offers and can perform. A service is composed of the interaction among software components in a system.

All components of set C should be deployed on m host nodes. Matrix CH describes how to deploy components on host nodes.

$$C_{1} \quad C_{2} \quad \cdots \quad C_{n}$$

$$H_{1} = \begin{pmatrix} ch_{1,1} & ch_{1,2} & \cdots & ch_{1,n} \\ ch_{2,1} & ch_{2,2} & \cdots & ch_{2,n} \\ \vdots & \vdots & \vdots & \vdots \\ H_{m} & ch_{m,1} & ch_{m,2} & \cdots & ch_{m,n} \end{bmatrix}$$

Each entry $ch_{i,j}$ in matrix CH may be 1 or 0. $ch_{i,j} = \begin{cases} 0, & \text{if component } C_j & \text{is not deployed on host node } H_i \\ 1, & \text{if component } C_j & \text{is deployed on host node } H_i \end{cases}$

For a system comprising m host nodes and n software components, the number of system deployment architectures is m^n . In general, determing the system deployment architecture that will maximize its reliability for the given target environment is an exponential complexity problem.

B. Frequency of Interacton among Software **Components**

Matrix CC describes frequency of interaction among software components. $C = \{C_1, C_2, \dots, C_n\}$ is a set of components. Each entry $cc_{i,j}$ in matrix CC represents the frequency of interaction between software component C_i and C_j . $C_{i,j}$ is an integer number in [a,b]. The values of a and b depend on concrete system.

$$C_{1} \quad C_{2} \quad \cdots \quad C_{n}$$

$$C_{1} \quad \begin{bmatrix} cc_{1,1} & cc_{1,2} & \cdots & cc_{1,n} \\ cc_{2,1} & cc_{2,2} & \cdots & cc_{2,n} \\ \vdots & \vdots & \vdots & \vdots \\ C_{n} \quad \begin{bmatrix} cc_{n,1} & cc_{n,2} & \cdots & cc_{n,n} \end{bmatrix}$$

III. COMPONENT RELIABILITY AND COMPONENT REPLICA

System reliability can be appropriately evaluated through component reliability. When there is not enough failure data, it is very common to make certain reasonable assumptions. A common assumption is that the system lifetime follows an exponential distribution. In this section, we suppose that all components follow exponential distribution and particularly depend on frequency of interaction among software components, as shown in formula (1). RC_i is the reliability of component C_i .Component failure rate λC_i is a function of interaction frequency fC_i of component C_i .

$$RC_i = e^{-\lambda C_i \times t}, i = 1, 2, \dots, n$$
 (1)

For a system consisting of m host nodes and nsoftware components, if component C_i has been identified as a candidate for replication, C_i should be replicated m-1 times. That is, C_i has m-1 component replicas or m components provide the same service. With redundant copies, a replicated component can continue to provide a service in spite of the failure of some of its copies.

$$\lambda C_{i} = \begin{cases} \lambda C_{i}^{"}, \text{ if } C_{i} \text{ is replicated} \\ \lambda C_{i}^{'}, \text{ otherwise} \end{cases}, i = 1, 2, \dots, n$$
(2)

$$\lambda C_i' = f C_i \times \rho_c + \lambda_0 + f_{other} \times \rho_{other}$$
(3)

$$\lambda C_i " = \lambda_0 + f_{other} \times \rho_{other} \tag{4}$$

- frequency criticality of interaction among software components.
- ρ_{other} criticality of other factors, it is a real number in [-1,1].
- λ_0 initial value of component failure probability.
- f_{other} influence of other factors on component reliability.
- fC_i frequency sum of interaction between component C_i and other components deployed on other host nodes

The value of fC_i can be obtained by matrix CC in formula(5). $H(C_i)$ describes the host node that component C_i is deployed on.

$$fC_i = \sum_{\forall C_j, \ H(C_i) \neq H(C_j)} cc_{i,j}$$
(5)

IV. SYSTEM RELIABILITY

It is often infeasible or difficult to directly estimate complex system reliability through large sample system-level test. Such difficulties may arise when the system-level test is costly or leads to destruction of the system itself. Nevertheless, system reliability can be appropriately evaluated through the component reliability information [14]. In this section, we suppose that the main sources of system failure are software component failure and host node failure. Therefore, system reliability is calculated in formula (6).

$$R_{system} = 1 - p_{system} = 1 - \bigcup_{i=1}^{m} (1 - ph_i) - \bigcup_{i=1}^{m} pch_i$$

$$pch_i = 1 - \prod_{\forall C_j, H(C_j) = H_i} ((1 - ph_i) \times e^{-\lambda C_i \times t})$$
(6)
$$(7)$$

 R_{system} system reliability

 p_{system} failure probability of system

 $H(C_j)$ the host node that C_j is deployed on

- pch_i failure probability of components deployed on host node H_i
- ph_i failure probability of host node H_i

V. EXPERIMENTS

In this section, four experiments are based on such a system consisting of eight original software components and four host nodes. Each original software component can provide a different service. The inputs of the experiments include randomly generated frequency matrix of interaction among software components and failure probabilities of four host nodes. Additionally, four experiments are based on three different deployment architectures.

A. Inputs

Matrix CC is randomly generated frequency matrix of interaction among software components. Each entry in matrix CC is an integer number [0, 7]. We can calculate the value of fC_i for component C_i on the basis of matrix CC .

	0	3	0	0	0	0	7	1]
	3	0	7	0	2	2	0	0
	0	7	0	6	0	5	0	0
C C	0	0	6	0	3	0	0	7
C C =	0	2	0	3	0	7	0	4
	0	2	5	0	7	0	3	0
	7	0	0	0	0	3	0	0
	1	0	0	7	4	0	7 0 0 0 0 3 0 0	0

Failure probabilities of four host nodes are real numbers in [0, 0.02]. ph_i represents the failure probability of host node H_i . $ph_1 = 0.0093$, $ph_2 = 0.0195$, $ph_3 = 0.0069$ $ph_4 = 0.0101$

B. Deployment Architecture and Frequency of Software Component Interaction

These eight components should be deployed on four host nodes. On the basis of frequency matrix of interaction among software components, we obtain three typical deployment architectures.

1) Deployment architecture

We use matrix CH_1 to describe the first deployment

architecture. CH_1 shows that C_4, C_5, C_8 are deployed on host node H_1, C_6, C_7 are deployed on H_2 . C_1, C_2 are deployed on H_3 . C_3 is deployed on H_4 .

$$CH_{1} = \begin{bmatrix} 0 & 0 & 0 & 1 & 1 & 0 & 0 & 1 \\ 0 & 0 & 0 & 0 & 0 & 1 & 1 & 0 \\ 1 & 1 & 0 & 0 & 0 & 0 & 0 & 0 \\ 0 & 0 & 1 & 0 & 0 & 0 & 0 & 0 \end{bmatrix}$$

We use matrix CH_2 to describe the second deployment architecture. Matrix CH_2 shows that C_8 is deployed on host node H_1 . C_5 , C_6 are deployed on H_2 . C_1 , C_7 are deployed on H_3 . C_2 , C_3 , C_4 are deployed on H_4 .

$$CH_2 = \begin{bmatrix} 0 & 0 & 0 & 0 & 0 & 0 & 0 & 1 \\ 0 & 0 & 0 & 0 & 1 & 1 & 0 & 0 \\ 1 & 0 & 0 & 0 & 0 & 0 & 1 & 0 \\ 0 & 1 & 1 & 1 & 0 & 0 & 0 & 0 \end{bmatrix}$$

We use matrix CH_3 to describe the third deployment architecture. CH_3 shows that $^{C_1, C_5}$ are deployed on host node H_1 . $^{C_4, C_6}$ are deployed on H_2 . $^{C_2, C_7}$ are deployed on H_3 . $^{C_3, C_8}$ are deployed on H_4 .

<i>CH</i> ₃ =	[1	0	0	0	1	0	0	0]
	0	0	0	1	0	1	0	0
$CH_3 =$	0	1	0	0	0	0	1	0
	0	0	1	0	0	0	0	1

2) Frequency of component interaction

On the basis of matrix CC, we calculate fC_i of each component C_i in formula (5). F_1 represents the set of fC_i of the first deployment architecture.

$$F_1 = \begin{bmatrix} 8 & 11 & 18 & 6 & 9 & 14 & 7 & 1 \end{bmatrix}$$

Therefore, it is easy to know that $fC_3 = 18$ and $fC_8 = 1$ in the first deployment architecture.

Similarly, F_2 represents the set of fC_i of the second deployment architecture. F_3 represents the set of fC_i of the third deployment architecture.

$$F_2 = \begin{bmatrix} 4 & 7 & 5 & 10 & 9 & 10 & 3 & 12 \end{bmatrix}$$
$$F_3 = \begin{bmatrix} 11 & 14 & 18 & 16 & 16 & 17 & 10 & 12 \end{bmatrix}$$

C. Experiments

The system includes eight software components. Each software component can provide a different service. The values of relevant parameters are $f_{other} = 0$ and $\rho_{other} = 0$.

1) Experiment one

In this experiment, $\lambda_0 = 0.0004$ and $\rho_c = 0.00001$. System reliability of three different deployment architectures can be shown in Fig.1. EFDSR represents system reliability of the first deployment architecture. ESDSR represents system reliability of the second deployment architecture. ETDSR represents system reliability of the third deployment architecture.



Figure 1. System reliability of three deployment architectures

As seen in Fig. 1, with the increasing of system run-time, system reliability of the second deployment architecture is obviously highest among the three. The impact of the first deployment architecture on system reliability is similar to the third one during [0, 70] hours.

A detailed analysis of component reliabilities of the three different deployment architectures is shown in Tab.I.

RH represents the highest of component reliability among the three deployment architectures. RM represents the medium value of component reliability

TABLE I. THE EXPERIMENTAL RESULTS FOR COMPONENT

	First	Second	Third	
	deployment	deployment	deployment	
	architecture	architecture	architecture	
<i>C</i> ₂	RM	RH	RL	
C_4	RH	RM	RL	
<i>C</i> ₅	RS	RS	RM	
C_8	RH	RS	RS	

among the three deployment architectures. RL represents the lowest of component reliability among the three deployment architectures. RS describes that there is a negligible difference of component reliability between two deployment architectures.

As seen in Tab.I, component reliabilities of component C_2 and C_4 of three deployment architectures are obviously different. There is a negligible difference

1143

of component reliabilities of C_5 and C_8 between two deployment architectures. Therefore, if C_4 is the most important component, we will deploy components on host nodes according to the second deployment architecture.

2) Experiment two

In this experiment, the first deployment architecture is the basis of the experiment. The value of ρ_c may be different. These values are $\rho_c = 0.00001$, $\rho_c = 0.00002$ and $\rho_c = 0.00004$. The influence of the values of ρ_c on system reliability can be shown in Fig. 2. With the higher value of ρ_c , system reliability becomes the lower. With the increasing of system run-time, the difference of system reliabilities with different values of ρ_c is more apparent.



Figure 2. System reliability of different value of ρ_c

The influence of the values of ρ_c on reliability of component C_3 is shown in Fig. 3.



Figure 3. Component C_3 reliability of different values of ρ_c

With the higher value of ρ_c , reliability of component C_3 becomes the lower. With the increasing of system run-time, the difference of reliability of

component C_3 of different values of ρ_c is more apparent.

3) Experiment three

In this experiment, we investigate the impact of system redeployment on system reliability. The value of ρ_c is 0.00004. Before thirty hours, system runs the third deployment architecture. At thirty hours, system reliability is less than 0.8. After thirty hours, we improve system reliability by redeployment, as shown in Fig. 4.



Figure 4. System reliability with system redeployment

After thirty hours, system begins to redeploy and run the second deployment architecture. In Fig.4, BDA represents system reliability of third deployment architecture during [0, 30] hours. BDE represents system reliability without system redeployment during [30, 80] hours. ADA represents system reliability with system redeployment during [30, 80] hours.

After system redeployment, component reliability has changed. Component reliability can be calculated in formula (8).

$$RC_{i} = \begin{cases} e^{-(\lambda_{0} + dfC_{i} \times \rho_{c}) \times t} & ,0 \le t \le 30\\ e^{-(\lambda_{0} + dfC_{i} \times \rho_{c}) \times t} & ,30 \le t \end{cases}, i = 1, 2, \dots, 8$$

$$(8)$$

 bfC_i represents interaction frequency of component C_i before redeployment. afC_i represents interaction frequency of component C_i after redeployment.

4) Experiment four

In this experiment, we investigate the difference of system reliability by replicating components and system redeployment. The value of ρ_c is 0.00004. When system reliability drops to some value, system reliability need to be improved by system redeployment or replicating components, as shown in Fig.5.



Figure 5. System reliability with redeployment and component replica

CC3 represents system reliability with replicating component C_3 during [30,80]hours. CC38 represents system reliability with replicating component C_3 and C_8 simultaneously during [30,80]hours. The meanings of BDA, BDE and ADA are illustrated in experiment three. As seen in Fig. 5, system reliability with replicating multiple components is higher than system redeployment. System reliability with redeployment is higher than replicating single component. System reliability will be improved by replicating components and system redeployment. However, if replicating components, we need to take into consideration the computational resources required and those available at each host node; if system redeployment, we need to consider the time to redeploy and the cost of system redeployment.

VI. CONCLUSIONS AND FUTURE RESEARCH

It is very important to estimate system reliability based on software architecture. Reliability is one of the most critical extra-functional properties of a software system. paper analyzes the defects of existing This architecture-level reliability estimation approaches, and proposes a novel system reliability estimation model incorporating the influence of component reliability, software deployment and component replica. Different deployment architectures have a significant influence on system reliability and component reliability. We present how to calculate system reliability and component reliability. We present the approaches of improving system reliability and the conditions of applying these approaches. In future research, system reliability estimation model based on SA will include other influence factors, such as software architectural styles, component replica strategies and so on.

ACKNOWLEDGMENT

The authors wish to thank Tao Liu for his kindly help with checking.

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Xihong Su was born in 1981 in China. She received the M.S. degree in school of computer science and technology from Harbin Institute of Technology, Harbin, China, in 2007. She is currently pursuing the Ph.D. degree at Harbin Institute of Technology. Her research interests include software architecture, software deployment and system reliability estimation.

Hongwei Liu was born in 1971 in China. He received the Ph.D. degree in school of computer science and technology from the Harbin Institute of Technology, PR China, in 2004. He is a professor of the School of Computer Science and Technology at the Harbin Institute of Technology. His main research interests are fault-tolerant computing, software reliability evaluation, software testing. He is a senior member of the CCF.

Zhibo Wu was born in 1947 in China. He received the Ph.D. degree in school of computer science and technology from Harbin Institute of Technology, Harbin, China, in 1980. He is a professor and Ph.D. supervisor in school of computer science and technology at Harbin Institute of Technology. His research interests include fault-tolerant computing, mobile computing and system reliability estimation. He is a senior member of the CCF.

Xiaozong Yang was born in 1939 in China. He received his B.S.degree in School of Computer and Science from Harbin Institute of Technology, in 1964. He is currently a professor and Ph.D. supervisor in School of Computer Science and Technology at Harbin Institute of Technology (CS HIT), the computer science institute director, the wearable computing Engineering and Research center Director of CS, HIT, vice director of fault tolerant computing committee under China computer federation and the head of wearable computing group under china computer Federation. His main research interests include computing architecture, fault tolerant computing. He has implemented many research projects and has received progress in his research work, and won about 10 awards of ministries. He published 2 books and about 100 papers, 10 PHD students graduated from his group, and was appointed as conference chair, session chair of China conference and international conference.

Decheng Zuo was born in 1972 in China. He received the Ph.D. degree in School of Computer and Science from Harbin Institute of Technology, in 2001. He is a professor of the School of Computer Science and Technology at the Harbin Institute of Technology. His main research interests include fault-tolerant computing, mobile computing and system reliability estimation. He is a senior member of the CCF.